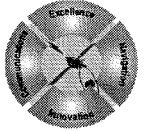


# **ADVANTAGES OF REGENERATIVE RANGING FOR DEEP SPACE NAVIGATION**

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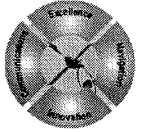
**DESCANSO Symposium**

9/23/99

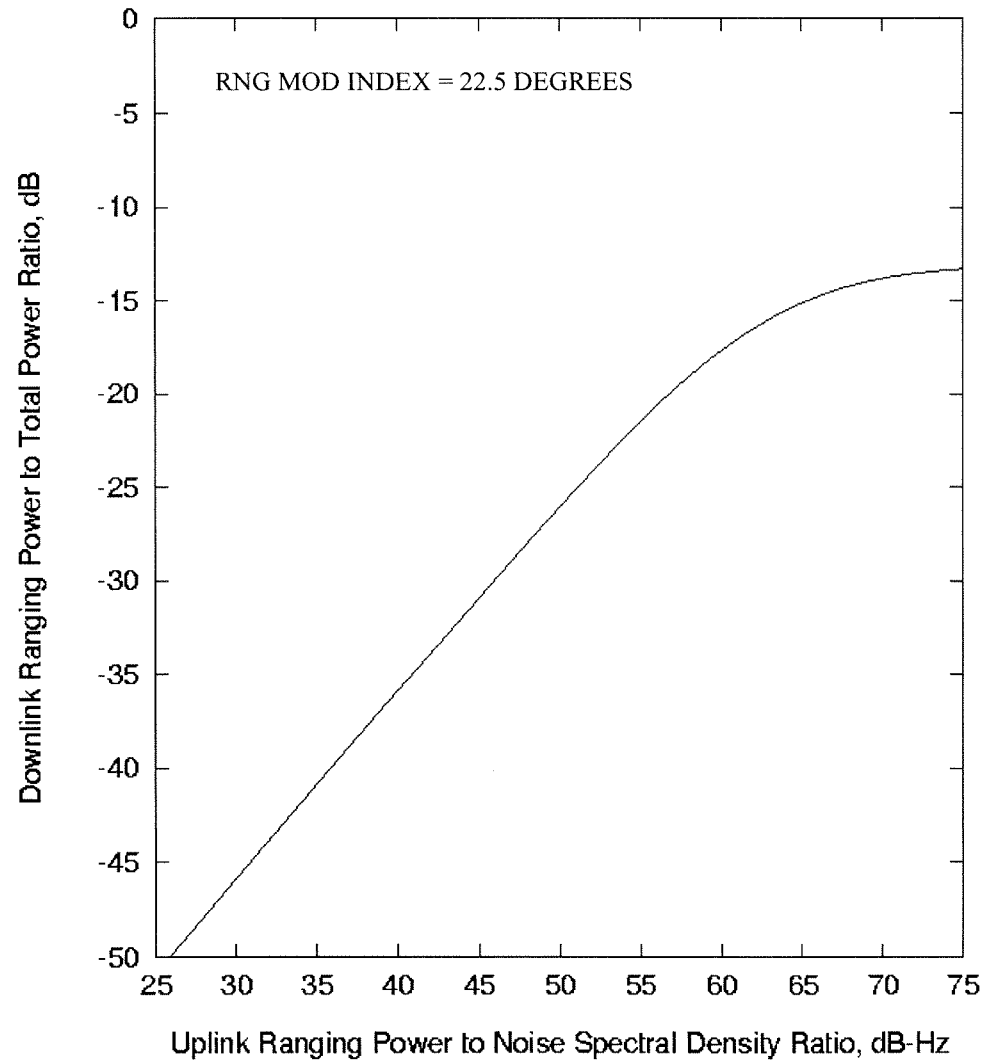


## CURRENT RANGING DESIGN

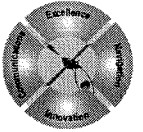
- RANGING SIGNAL IS MODULATED ONTO THE UPLINK
- SPACECRAFT DEMODULATES THE RANGING SIGNAL AND FILTERS IT WITH A 1.5 MHz FILTER
  - 1.5 MHz OF NOISE IS INCLUDED WITH THE SIGNAL
- DOWNLINK CARRIER IS MODULATED BY THE FILTERED RANGING SIGNAL
  - NOISE DEGRADES THE DOWNLINK
- GROUND EQUIPMENT DEMODULATES THE RANGING SIGNAL FROM THE CARRIER AND CORRELATES IT WITH THE REFERENCE
  - INTEGRATION TIME IS INCREASED TO REDUCE THE NOISE



## TURN AROUND DEGRADATION

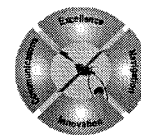


DESCANSO Symposium



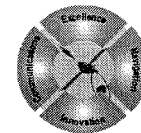
## SEQUENTIAL RANGING

- **SERIES OF SQUARE WAVE TONES SENT**
  - FREQUENCY OF EACH TONE HALF THE FREQUENCY OF THE PREVIOUS ONE
  - HIGHEST FREQUENCY TONE CALLED THE CLOCK COMPONENT
  - TOTAL NUMBER OF COMPONENTS (N) DEPENDS ON THE AMBIGUITY TO BE RESOLVED
- **CLOCK SENT FOR PERIOD OF TIME (T1 SECONDS)**
  - THEN EACH COMPONENT SENT FOR T2 SECONDS
- **CLOCK SENT AGAIN FOR DRVID (DIFFERENCED RANGE VS INTEGRATED DOPPLER) MEASUREMENTS**
  - T3 SECONDS FOR EACH DRVID MEASUREMENT
  - NDRVID MEASUREMENTS
- **TOTAL TIME FOR 1 RANGE MEASUREMENT (CYCLE TIME) IS**
  - $(2+T1) + (1+T2)*(N-1) + NDRVID*(2+T3) + 1$
- **SINCE EACH COMPONENT MUST BE CORRELATED AND INTEGRATED SEPARATELY, ALL ACQUISITIONS MUST START AT BEGINNING OF SEQUENCE**

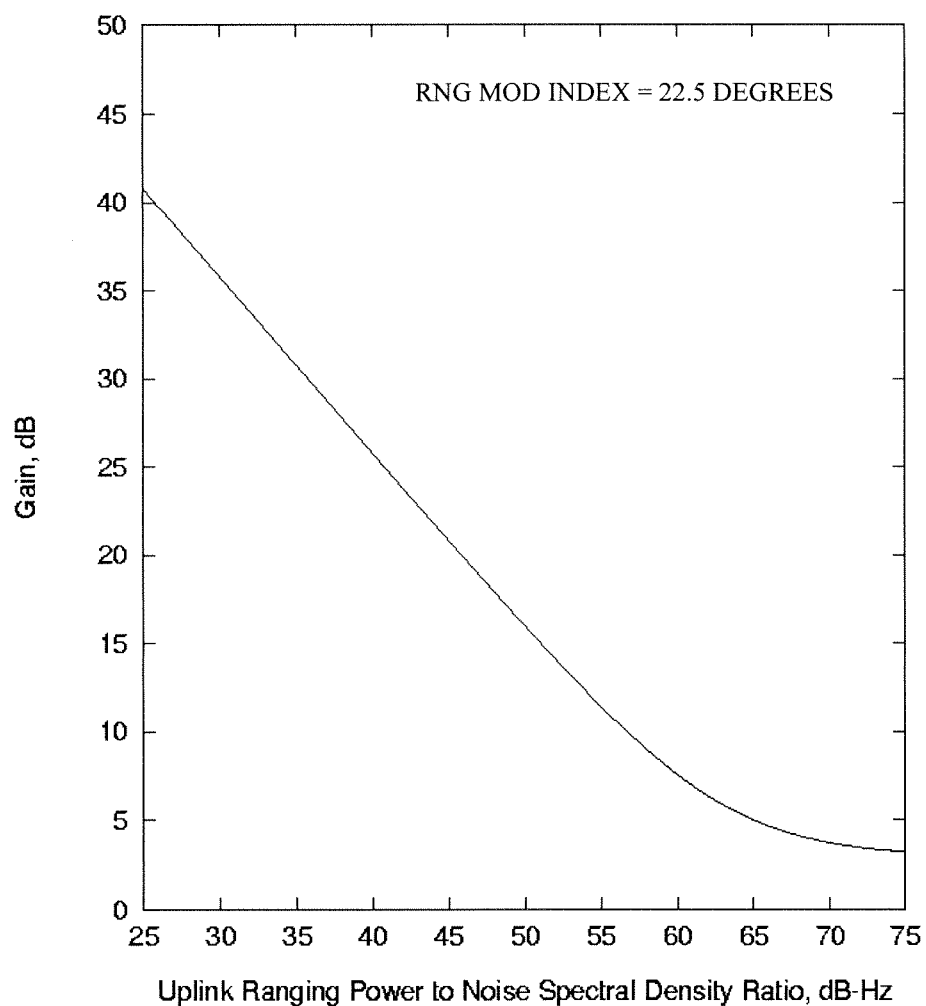


## REGENERATIVE RANGING

- **INSTEAD OF FILTERING THE RECEIVED SIGNAL PLUS NOISE, THE RANGING SIGNAL IS TRACKED ON THE SPACECRAFT**
- **ONCE THE SPACECRAFT LOCKS TO THE RANGING SIGNAL, THE REGENERATED SIGNAL IS USED TO MODULATE THE DOWNLINK CARRIER**
  - **THIS ALLOWS SIGNIFICANT REDUCTION IN THE NOISE ON THE SIGNAL, FROM 1.5 MHz TO THE SIGNAL TRACKING LOOP BANDWIDTH**
- **THIS REQUIRES THAT THE SPACECRAFT HAVE SOME KNOWLEDGE OF WHAT THE RANGING SIGNAL LOOKS LIKE AND THAT THERE IS ENOUGH SNR TO LOCK**
  - **SEQUENTIAL SIGNAL IS NOT A GOOD CANDIDATE, DUE TO LONG CYCLE TIMES AND VARIABLE PARAMETERS**



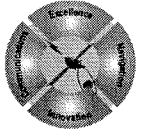
## REGENERATIVE RANGING GAIN





## **PN CODE DESCRIPTION**

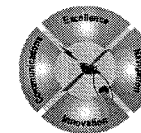
- **PSEUDO-NOISE (PN) RANGING PROVIDES AN EASIER SIGNAL TO LOCK TO**
- **PN SIGNAL IS GENERATED BY OR'ING THE CLOCK COMPONENT WITH THE RESULT OF AND'ING 5 PN SEQUENCES**
  - **SEQUENCES OF LENGTH 7, 11, 15, 19, 23**
  - **RESULTING SEQUENCE IS 1,009,470 CHIPS LONG**
  - **CLOCK RATE IS APPROXIMATELY 1 MHz**
    - **RESULTING CYCLE TIME IS 0.5 SECONDS (2 CHIPS PER CLOCK CYCLE)**
    - **CLOCK IS FREQUENCY COHERENT WITH THE UPLINK CARRIER**
- **PN SIGNAL LOOKS LIKE A SQUARE WAVE AT 1 MHz, WITH A FEW ERRORS**



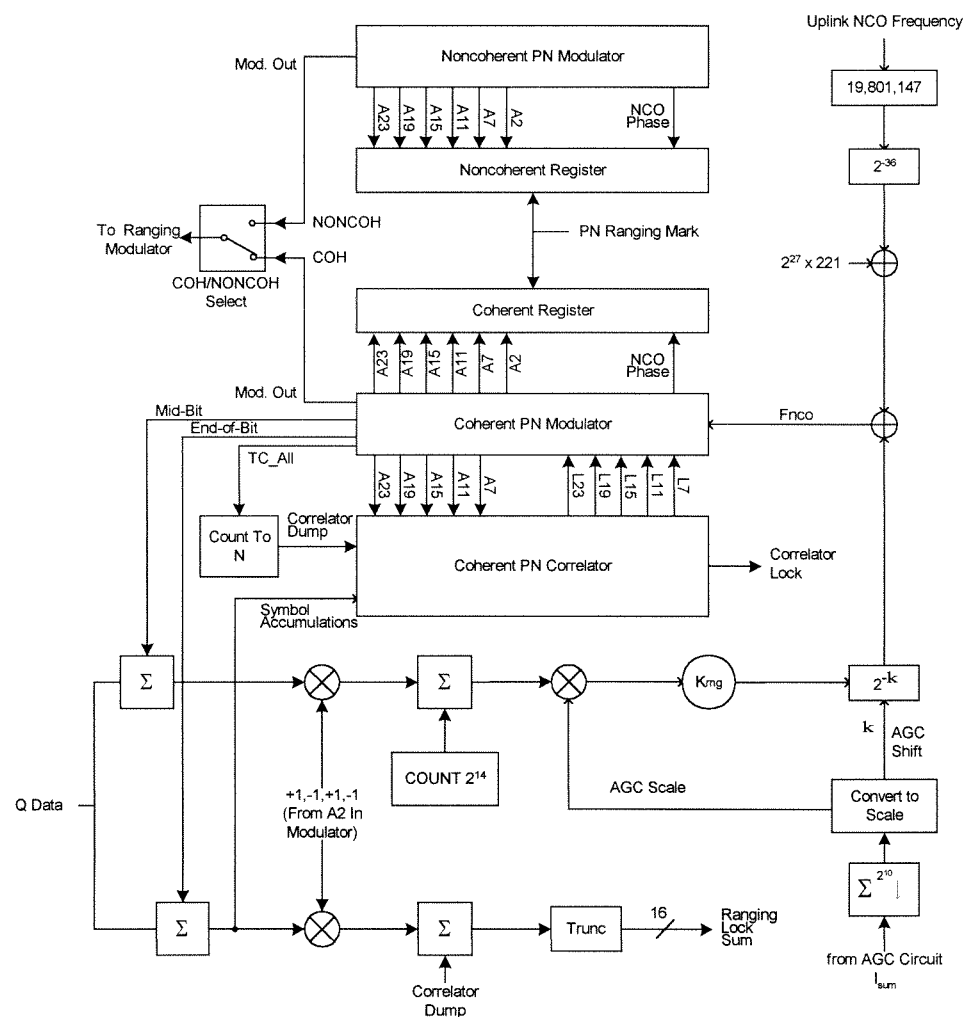
## STM DESIGN

- THE SPACECRAFT TRANSPONDING MODEM (STM) HAS IMPLEMENTED THE REGENERATIVE CIRCUITRY
- THE CARRIER FREQUENCY IS SCALED TO GET THE CLOCK FREQUENCY
  - SINCE ONLY PHASE NEEDS TO BE TRACKED, A FIRST ORDER PHASE LOCKED LOOP IS USED TO TRACK THE CLOCK (CHIP) PHASE
- ONCE THE CHIPS ARE IN LOCK, THE SIGNAL IS CORRELATED AGAINST THE PN SEQUENCE
  - CORRELATIONS ARE DONE AGAINST THE SUBSEQUENCES, SO ONLY 75 CORRELATIONS ARE NEEDED
  - FOR WORST CASE SIGNAL STRENGTH, ONLY 18 SECONDS INTEGRATION TIME NEEDED
- WHEN OFFSET IS DETERMINED, PN SEQUENCE USED TO MODULATE DOWNLINK CARRIER





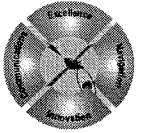
## REGENERATIVE CIRCUIT





## **USES OF IMPROVEMENT**

- **THE GAIN IN RECEIVED RANGING POWER CAN BE USED IN SEVERAL WAYS:**
  - **REDUCE THE INTEGRATION TIME ON THE GROUND (TO REDUCE THE TRACKING TIME DEVOTED TO RANGING), GIVING THE SAME MEASUREMENT VARIANCE**
  - **KEEP THE INTEGRATION TIME ON THE GROUND THE SAME, REDUCING THE VARIANCE ON THE MEASUREMENT**
  - **REDUCE THE DOWNLINK RANGING MODULATION INDEX, KEEPING THE INTEGRATION TIME ON THE GROUND THE SAME, PROVIDING MORE POWER FOR TELEMETRY**



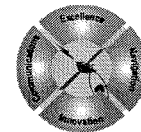
## MARS GLOBAL SURVEYOR (MGS) EXAMPLE

### CURRENT

NUMBER OF COMPONENTS	18
NUMBER OF DRVID	0
T1 (SEC)	20
T2 (SEC)	20
T3 (SEC)	0
CYCLE TIME (SEC)	210
UPLINK Pr/N0 (dB-Hz)	44.9
DOWNLINK Pr/N0 (dB-Hz)	10.0

### WITH REGENERATION

SPACECRAFT INTEGRATION TIME (SEC) (ASSUMING 0.999 PROB. OF ACQ.)	0.5
DOWNLINK Pr/N0 GAIN (dB)	20.6



## **CASSINI EXAMPLE**

<b><u>CURRENT</u></b>	<b><u>VENUS CRUISE</u></b>	<b><u>SATURN</u></b>
<b>NUMBER OF COMPONENTS</b>	<b>10</b>	<b>TBD</b>
<b>NUMBER OF DRVID</b>	<b>0</b>	<b>TBD</b>
<b>T1 (SEC)</b>	<b>1248</b>	<b>TBD</b>
<b>T2 (SEC)</b>	<b>60</b>	<b>TBD</b>
<b>T3 (SEC)</b>	<b>0</b>	<b>TBD</b>
<b>CYCLE TIME (SEC)</b>	<b>1800</b>	<b>TBD</b>
<b>UPLINK Pr/N0 (dB-Hz)</b>	<b>35.8</b>	<b>61.8</b>
<b>DOWNLINK Pr/N0 (dB-Hz)</b>	<b>-8.4</b>	<b>12.0</b>
<b><u>WITH REGENERATION</u></b>		
<b>SPACECRAFT INTEGRATION TIME (SEC) (ASSUMING 0.999 PROB. OF ACQ.)</b>	<b>2.3</b>	<b>0.5</b>
<b>DOWNLINK Pr/N0 GAIN (dB)</b>	<b>27.1</b>	<b>6.3</b>